test wolf becoperations and Proposed Flow Test LH_2

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Introduction: Approaches to Heat Removal

Two approaches under consideration:

- External cooling loop (traditional approach).
- ${}^{ imes}$ Bring the LH_2 to the coolant (heat removed in an external heat exchanger).
- Combined absorber and heat exchanger.
- absorber). It is the LH_2 (remove heat directly within absorber).





- Advantages/disadvantages of an external cooling loop:
- .(8213 DAJ2 .g.s) ziargets (e.g. SLAC E158). $+\,$
- $+~{
 m Easy}$ to regulate bulk temperature of $LH_2.$
- + Is likely to work best for small aspect ratio (L/R) absorbers.
- May be difficult to maintain uniform vertical flow through the absorber.

Advantages/disadvantages of a combined absorber/heat exchanger:

- + Takes advantage of natural convection transverse to the beam path.
- + Flow in absorber is self regulating, *i.e.* larger heat input \Rightarrow more turbulence \Rightarrow enhanced thermal mixing.
- + Is likely to work best for large aspect ratio (L/R) absorbers.
- More difficult to ensure against boiling at very high Rayleigh numbers.



Energy balance between LH_2 and coolant ($H\epsilon$).

 $d_{\mathcal{J}}$

=

thermal conductivity of cooling tube walls	_ =	$^{m}\mathcal{Y}$	
thickness of cooling tube walls	. =	$x_{ abla}$	
convective heat transfer coefficient of $H\epsilon$	=	${}^{\scriptscriptstyle {artheta}}H \mathcal{U}$	
convective heat transfer coefficient of LH_2	=	$\gamma^{TH^{7}}$	
surface area of cooling tubes	=	abla	
bulk temperature of LH_2	=	$L^{TH^{5}}$	
coolant outlet temperature	=	^{o}L	
coolant inlet temperature	=	L^{i}	
	:S19	Parameto	<u> </u>

specific heat capacity of He



Heat Exchanger Analysis (cont'd)

Rate of heat transfer:

$$\frac{(\overline{i}T - \underline{c}^{H}L^{T})}{(\overline{i}T - \underline{c}^{H}L^{T})} \prod \left(\frac{\underline{1}}{\underline{o}^{H}L} + \frac{\underline{x}\Delta}{\underline{w}} + \frac{\underline{1}}{\underline{c}^{H}L}\right)}{(\underline{i}T - \underline{o}^{H}L^{T})} = \dot{P}$$

:9H To star volt as He:

$$\frac{\dot{V}}{C_p \left(T_o - vT\right)} = {}_{oH}\dot{m}$$

 $h_{H_c} \Rightarrow$ from appropriate correlation (flow through a tube). h_{LH_2} and $T_{LH_2} \Rightarrow$ from CFD simulations (no



Computational Fluid Dynamics (CFD)

Features of the CFD Simulations:

- \checkmark Provides average convective heat transfer coefficient and average LH_2 temperature for heat exchanger analysis.
- \checkmark Track maximum LH_2 temperature (cf. boiling point).
- Determine details of fluid flow and heat transfer in absorber.
- \Rightarrow Better understanding leads to better design!



- Take 1: Results using FLUENT (M. Boghosian):
- ✓ Simulate one half of symmetric domain.
- Steady flow calculations.
- ✓ Heat generation via steady Gaussian distribution.
- .⁸01 × $h \leq _{A} n A$ rot besu (SNAA) gnileborn econemutint \checkmark
- Take 2: Results using COA code (A. Obabko and E. Almasri):
- . Simulate full domain.
- Unsteady flow calculations.
- All scales computed for all Rayleigh numbers.
- $-x_{nm}T$ ni tookarev qutup versior, e.g. startup overshoot in T_{max} .

Investigate possibility of asymmetric flow oscillations.

Investigate influence of beam pulsing.



FLUENT CFD Results







Ъа_р

FLUENT CFD Results (cont'd)

Non-Dimensional Maximum Temperature vs. Rayleigh Number:



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Parameter Map for LH_2





Note: Properties taken at 18 K, 2 atm.

COA Formulation

Properties and parameters:

0 = thermal diffusivity of	al diffusivity of LH_2
V = kinematic viscosity c	atic viscosity of LH_2
\dot{q}' = rate of heat generati	heat generation per unit length
$\dot{q}^{\prime\prime\prime}(r)$ = rate of volumetric he	volumetric heat generation (Gaussian distribution)
: Two states is the second state of the second states of the second states T_{m}	mperature of absorber
= radius of absorber	of absorber





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Energy equation:

$$(1) + u_r \frac{\partial U}{\partial L} + u_r \frac{\partial \theta}{\partial L} = \frac{\partial \theta}{\partial \theta} \frac{\partial U}{\partial L} = \frac{\partial U}{\partial L} + \frac{U}{1} \frac{\partial U}{\partial L} + \frac{U}{1} \frac{\partial U}{\partial L} + \frac{U}{1} \frac{\partial \theta}{\partial L} + \frac{U}$$

Vorticity-transport equation:

$$\frac{\partial\omega}{\partial t} + v_r \frac{\partial\omega}{\partial r} + \frac{v_\theta}{r} \frac{\partial\omega}{\partial \omega} = P_r \left[\frac{\partial^2\omega}{\partial r^2} + \frac{1}{r} \frac{\partial\omega}{\partial r} + \frac{1}{r^2} \frac{\partial\omega}{\partial r} + \frac{1}{r^2} \frac{\partial\omega}{\partial \theta^2} \right]$$
$$+ Ra_R P_r \left[\sin\theta \frac{\partial\Gamma}{\partial r} + \frac{\partial\Omega}{r} + \frac{\partial\Omega}{r} + \frac{\partial\Omega}{r} \right]$$

Streamfunction equation:

$$u_{r} = \frac{r}{r} \frac{\partial \theta}{\partial v}, \quad u_{\theta} = -\frac{\partial \eta}{\partial v}, \quad u_{\theta} = -\frac{\partial \eta}{\partial v}$$
$$u_{r} = \frac{1}{r} \frac{\partial \psi}{\partial v}, \quad u_{\theta} = -\frac{\partial \eta}{\partial v}$$



Formulation (cont'd)

Initial and boundary conditions:

$$0=t$$
 is $0=u_{ heta}=v_{ heta}=0$ at $t=0,$

. $1 = \tau$ is $0 = \theta_{\theta} = \tau = \psi = T$

$$I = \frac{\dot{q}'/\dot{k}}{L_* - L^m}, \quad \dot{\eta} = \frac{\dot{\alpha}}{\dot{\eta}_*}, \quad \eta = \frac{\dot{\alpha}}{\dot{\eta}_*}, \quad \eta = \frac{\dot{\alpha}}{\dot{\eta}_*}, \quad \chi = \frac{\dot{\alpha}/\dot{B}_2}{\dot{\eta}_*},$$
$$I = \frac{\dot{H}}{\dot{L}_*}, \quad v_{\theta} = \frac{\dot{H}/\dot{\alpha}}{\dot{\eta}_*}, \quad v_{\theta} = \frac{\dot{H}/\dot{\alpha}}{\dot{\eta}_*}, \quad t = \frac{\dot{H}_2/\dot{\alpha}}{\dot{\eta}_*},$$

$$q(r) = \frac{\dot{q}'/R^2}{\dot{q}''(r)}, \quad \psi = \frac{2\pi\sigma^2}{\psi^*}, \quad \omega = \frac{\alpha}{\omega'}, \quad \omega = \frac{R}{\omega'}.$$
$$q(r) = \frac{\dot{q}'/R^2}{\dot{q}''(r)}, \quad \psi = \frac{\omega}{\psi^*}, \quad \omega = \frac{\omega/R^2}{\omega'}.$$

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Formulation – Non-Dimensional Parameters

Prandtl Number:

$$\frac{\partial}{\partial t} = \lambda d$$

Rayleigh Number:

$$\mathcal{H}^{qB} = \mathcal{C}^{L} \mathcal{D}^{L} = \frac{\partial \mathcal{H}_{3}}{\partial \mathcal{H}_{3}} \frac{\partial \mathcal{H}_{4}}{\partial \mathcal{H}} \left(= \frac{35}{\pi} \mathcal{H}^{qMB} \right)$$

:neselt number:



- The following flow regimes are observed:
- ⁸⁰¹ Steady, symmetric solutions: $Ra_R \leq 1 \times 10^8$
- $^{0}01\times1\leq_{A}bA$:asymmetric solutions: $Ra_{R}\geq1\times10^{9}$

<u>6.0</u>

G.0-

<u>-0،5</u>

<u>5.</u>0

:(noiform heat generation) $^{7}01 \times ^{7}d.1 = _{R} nR$ (uniform heat generation):

Streamfunction: Temperature:



Vorticity:



Steady, Symmetric Results (cont'd)

:(noiform heat generation) $^701 \times 73.1 = _{R} n R$ for θ such a subset of the set generation):

:heta 'sa n_N





(uN) rede Comparisons – Average Nusselt Number (Nu)

. Uniform heat generation ($\sigma
ightarrow \infty$) with $P_T=1$:

15.0	6.11	0.41	$^{7}01 \times 73.1$
8.2	7.7	83.8	1.57×10^6
9boD AOD	EΓ∩EΝ⊥ ₃	Mitachi et al. ¹	Rak

¹ Mitachi et al. (1986, 1987) - Results shown are from numerical simulations which compared favorably with experiments.

. From M. Boghosian's correlation for Pr=1.4, *i.e.* \overline{N} and $NR=0.7041 \cdot Ra_{MB}^{0.1864}$.



Gaussian heat generation: $\sigma=0.25$

steady laminar, steady RANS (turbulent), unsteady N-S

97	0200.0	0.0039	3.85	0900.0	£₽00.0	$0^{1}0^{1} \times 1$
₽.85.4	110.0	2900.0	55.1	1010.0	2900.0	$^{001} \times 1$
9. ट 1	810.0	0.0100	₽. 9 I	6910.0	1010.0	$^{8}01 \times 1$
$n_{\underline{N}}$	L^{xpuu}	L	n <u>N</u>	L^{xpuu}	L	Rak
9boJ AOJ		ELUENT ¹				

: From M. Boghosian's correlations ($T_{MB} = \frac{\pi}{4}T$):

 $T_{avgMB} = 0.3130 \cdot R_{a} \frac{0.1771}{MB}, \quad T_{maxMB} = 1.3597 \cdot R_{a} \frac{0.2233}{MB}, \quad \overline{N}u_{MB} = 0.7041 \cdot R_{a} \frac{0.1852}{MB}, \quad \overline{N}u_{MB} = 0.7041 \cdot R_{a} \frac{0.1852}{MB}$



Steady, Symmetric Results: $Ra_R = 1 \times 10^8$, $\sigma = 0.25$









Steady, Symmetric Results: $Ra_R = 1 \times 10^8$, $\sigma = 0.25$





$\overline{\mathrm{d} \mathfrak{L}.0} = \mathfrak{O}, \overline{\mathrm{d} \mathfrak{I}01} imes 1 = R \mathfrak{B} R$:stlussR cirts Results: $R \mathfrak{B} R$

:heta 'sa n_N





$dS.0 = \overline{\sigma, 0^1}01 \times 1 = AbA$:słluseA symmetric Results: $Ra_R = 1 \times 10^{10}, \sigma = 0.25$





For dE/dx = 13.81 M, 1.5×10^{14} muons/s $\Rightarrow \dot{q}' = 332$ W/m. Then at 100 atm and 80 K $\Rightarrow Ra_R = 2.01 \times 10^{15}$ for R = 0.5 m.

Characteristics:

- !pniliod oN +
- More complex and time-consuming to solve the fluid flow and heat
- hydrogen absorber. It is one order of magnitude higher than in the case of liquid \mathbb{R}^R is one order.
- Sompressibility?
- ? Treatment of actual geometry.
- ? Effect of ionization and magnetic field on fluid flow and heat transfer characteristics.



Conclusions

- Current COA results compare very well with limited experimental data and FLUENT results (both laminar and turbulent regimes).
- \blacktriangleright Critical Rayleigh number for unsteady, asymmetric behavior is $Ra_R>1\times10^8$.
- results. The sponds to laminar to turbulent transition in FLUENT \Rightarrow Roughly corresponds to laminar to turbulent transition in FLUENT.
- ...M Abid ts start-up overshoot in temperature at high Ra.
- heat exchanger. \Rightarrow Heater not necessary to improve performance of absorber as
- CFD results offer guidance for gaseous absorber (additional issues



:tsil dsiW

- \checkmark Near room temperature flow test \Rightarrow minimize cost; maximize possible sites for test.
- ✓ Working fluid that is safe and easy to work with.
- Allow for flexibility in providing heat source.
- Maximize information obtained without need for internal measurements (may be difficult depending on heat source).
- \Rightarrow If such measurements are possible, all the better.
- Provide for comparisons of essential data with CFD results.



In a typical test one would choose the geometry, working fluid and heat input to give a particular Rayleigh number. Then the temperature (e.g. maximum temperature) and flow conditions would be measured.

.* $T\Delta$ ənimrəfəb bns rədmun dgiəl χ sЯ əhf əzooh $\Im \leftarrow$

The key insight:

temperature to boiling, i.e. $\Delta T^* = T^*_{boil} - T^*_w$.

range are chosen, and the required heat input is determined.

 \Rightarrow Choose the ΔT^* and determine Rayleigh number.



Features:

- Set up: absorber encased in cooling sheath (similar to actual absorber).
- Heat source: electric current in absorber fluid, beam, etc.
- Absorber fluid: water is a candidate.
- Could possibly use additive to increase electrical conductivity and/or



Parameter Map for Water



Note: Properties taken at 100°C, 1 atm.



Proposed Flow Test (cont'd)

Procedure:

- . The choose $\Delta T^* \Rightarrow Absorber wall temperature <math>T^*_w = T^*_{mod} = T^*$.
- (2) Circulate coolant until absorber fluid reaches uniform temperature equal to \overline{V}_w^* .
- Iurn on heat source and increase in a quasi-steady manner, i.e. slowly, Inti incipient boiling occurs.
- Acord video to note location of incipient boiling and visualize flow using
 Dubbles.
- Determine heat output from absorber by measuring mass flow rate and inlet/outlet temperatures of coolant.
- (6) At conclusion of test, drain fluid from absorber and determine bulk, i.e. average, temperature, \overline{T}^*_{avg} , of absorber fluid, i.e. drain at constant, known mass flow rate and measure time series of temperature of draining fluid.
- .s'* $T\Delta$ for a series of ΔT *'s.



Shalysis of flow-test results:

- The Determine actual Rayleigh number of test from magnitude of heat input necessary to produce boiling, i.e. selected $\Delta T^* = {}^* T_{boil} {}^* T_w^*$.
- O Determine heat input predicted from CFD to produce temperature rise corresponding to ΔT^* .
- $^{\odot}$ Compare actual heat input required for boiling with that predicted from CFD, i.e. compare actual and predicted Rayleigh numbers for given $\Delta T^{*}_{.}$
- Betimate average Nusselt number using actual heat input, heat transfer \mathbb{G} . Estimate area and $\overline{\Gamma}^*_{wn} \overline{\Gamma}^*_{wn}$
- D Compare estimated Nusselt number from flow test with predicted value from CFD.



Features of flow-test:

- \checkmark Choose ΔT^* rather than Rayleigh number for each test.
- ✓ No exotic fluid flow or temperature measurements necessary.
- → We measure the maximum temperature visually by heating until boiling occurs.
- \Rightarrow The fluid may be heated in the most practical manner without \Rightarrow The fluid may be heated in the most practical manner without
- The bubbles provide some limited visualization capability.
- Measuring maximum temperature is a quantity that is influenced strongly by both fluid dynamic and heat transfer aspects, i.e. it is a composite of the entire fluid flow and heat transfer environment.

